ONE TIME EXIT SCHEME

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Seventh Semester B.E. Degree Examination, April 2018 **Gas Turbine Technology**

Ti	me:	3 hrs. Max. N	1arks:100
N	ote:	Answer any FIVE full questions, selecting atleast TWO questions from	each part.
		$\underline{PART} - \underline{A}$	
1	a. b. c.	components of turbo jet engine with after burner. Show that turbo fan is more efficient than turbo jet engine.	in various (10 Marks) (05 Marks) (05 Marks)
2	a. b. c.	Write the requirements of good thrust reverser.	(12 Marks) (04 Marks) (04 Marks)
3	a. b.	Explain in detail any five surface finishing processes. Explain any five methods of casting.	(10 Marks) (10 Marks)
4	a.	With a neat sketch, draw a typical fuel system and write function of its componer	nts.
	b. c.	The state of the state of the a near sketch.	(10 Marks) (06 Marks) (04 Marks)
		<u>PART – B</u>	
5	a. b. c.	Explain about Manned Aero thrust engine restart envelopes with neat sketch. Write the fundamental start phases of a engine. Define design, off design and transient performance.	(10 Marks) (05 Marks) (05 Marks)
6	a. b.	Explain about Rotating and Locked stall, with neat sketch. Explain about Axial flow turbine map.	(10 Marks) (10 Marks)
7	a. b.	The following conditions are known about a running engine, RPM = 9465 , EGT W_f = 1814.4 kg/hr , W_a = 90.7 kg/s , F_n = 4536kg , Barometric pressure = 1 Ambient temperature = 27.8°C. Calculate corrected above engine performance pathe standard conditions of 101.3KPa and 15°C. Explain about various preliminary flight ratings.	02.6 KPa
8	a. b. c.	Explain the measurement of thrust and shaft speed. Write the steps involved in test bed calibration. Write the purpose of pressure measurement.	(10 Marks) (06 Marks) (04 Marks)